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Comment on Flow through Multiple Shock Systems Edit. In example 4.12 (p. 155, Anderson), one case uses two oblique shocks (one incident and one reflected) to change the Mach number from an upstream value,  $M_1$ , of 2.8 to a downstream Mach number,  $M_3$ , of 1.45.

### **Modern Compressible Flow Chapter 4 Notes | Aero ...**

The inside surface of the tube is assumed a slip surface. Since the time span of the unsteady flow is short, the waves never reach the end walls, and so, conditions at these boundaries are held fixed. Comparison Data. The classical shock tube solution provides data for comparison. The text by Anderson discusses this solution.

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Modern Compressible Flow with Historical Perspective, Edition 3 (but older versions are acceptable too) John D. Anderson, McGraw-Hill, Inc., New York, 2003. Other notes and articles will be provided by the instructor. Comments: Here is a link to a website that carries this book for a very reasonable price: Buy online .

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